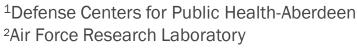


Aircrew Multi-Axis Vibration Exposures During Operation of the Blackhawk UH-60L Helicopter

Steven G. Chervak¹ Suzanne D. Smith, PhD² March, 2023







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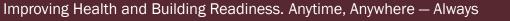




Background

- Military aircrews continue to report back discomfort, pain, and injury associated with flying rotary-wing aircraft.
- The Defense Centers for Public Health Aberdeen (DCPH-A) and USAF are collaborating on a project to expand limited data on aircrew operational vibration exposure.
- The project is funded by the National Defense Center for Energy and Environment, Safety & Occupational Health Focus Group.
- It focuses on four platforms in addition to an initial study conducted on the HH-60M and UH-72.
- This presentation focuses on the UH-60L Blackhawk.







Objectives

Characterize and assess aircrew vibration exposure aboard the UH-60L.

- Investigate multi-axis acceleration spectra for targeted flight test conditions.
- Apply MIL-STD 1472/ACGIH (ISO 2631-1) to assess comfort and health risk.
- Conduct an aircrew survey regarding discomfort, vibration, and equipment.
- Document data in the AFRL Collaborative Biomechanics Data Network (CBDN).





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Methods – Data Collection Unit

- Remote Vibration Environment Recorder (REVER)
- Portable and Battery-operated
- Four systems required for flight test

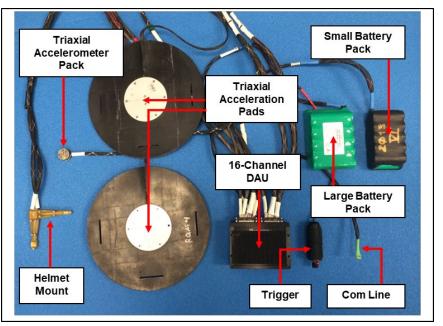


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Methods - Seat Locations

Station	Measurement Site	Sensor Type		
	Seat Base	Triaxial Accelerometer Pack		
Dilet (right easkpit east)	Seat Pan	Triaxial Acceleration Pad		
Pilot (right cockpit seat)	Seat Back	Triaxial Acceleration Pad		
	Helmet	Six-Axis Helmet Mount		
Crew Chief/Flight Engineer (mid cabin,	Floor beneath Seat	Triaxial Accelerometer Pack		
side-facing right seat)	Seat Pan	Triaxial Acceleration Pad		
	Seat Back	Triaxial Acceleration Pad		
Crew Chief/Flight Engineer (mid cabin,	Floor beneath Seat	Triaxial Accelerometer Pack		
side-facing left seat)	Seat Pan	Triaxial Acceleration Pad		
	Seat Back	Triaxial Acceleration Pad		
Crew Member (aft cabin, rear-facing	Floor beneath Seat	Triaxial Accelerometer Pack		
right seat)	Seat Pan	Triaxial Acceleration Pad		
Crew Member (aft cabin, rear-facing	Floor beneath Seat	Triaxial Accelerometer Pack		
left seat)	Seat Pan	Triaxial Acceleration Pad		



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Methods - Vest and Helmet Setup

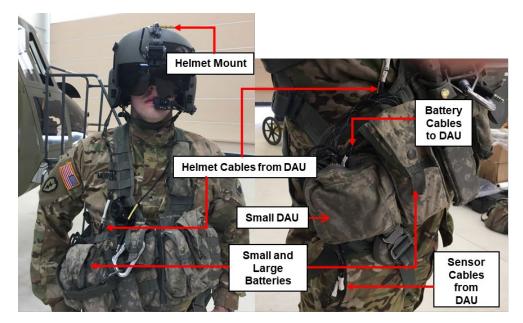


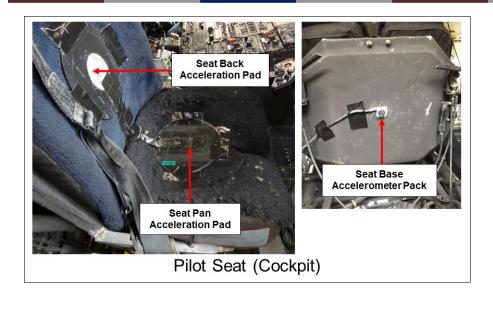
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Methods - Crew Seat Setup



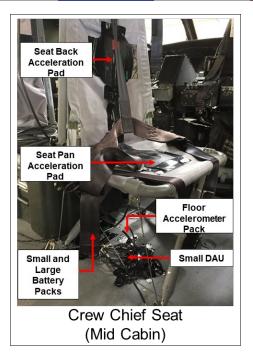


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Methods - Rear Seat Setup

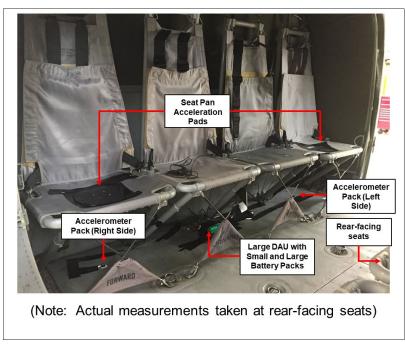


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Methods – Test Conditions

- Flight Test Conditions
 - Per Task
 - Multiple Records
- **Data Collection**
 - Acceleration time histories collected for 20 sec. for each condition upon trigger activation

FLIGHT TEST CARD			FLIGHT TEST CARD					
AC/#: LOCATION/DATE:			AC/#: LOCATION/DATE:					
PI: CP:		CC:		PI:	CP:		CC:	
Other:				Other:				
Flight #: Station:			Flight #: Station:					
CONDITION	ALT (ft		COMMENTS (Wind,		CONDITION	ALT (ft		COMMENTS (Wind,
(*Multiple Test Records Desired) MSL) (KCAS) Day, Night, etc.)			(*Multiple Test Records Desired) MSL) (KCAS) Day, Night, etc.)					
TASK 1024 Before Starting Through Before Leaving Helo Checks A. Engine Idle 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0			TASK 1024 Before Starting Through Before Leaving Helo Checks P. Steep Rate Turn 					
A. Engine lale Record #:		0		P. Steep Rat Record#:	eium	<10K	5120	
B. Ground Flight 100%	0	0		Q. Descent			≤120	
Record #:		0		Record #:			5120	
	040 Perform VI	C Takaa	"	R. AVCS Off		<10K	≤120	
C. Takeoff Normal	1040 Perform VN	A/R	<u>π</u>	Record #:		<1UK	5120	
C. Taken worman Aver Aver Aver Aver Aver Aver Aver Aver							a ch	
D. Takeoff Vertical	_			E Normal A	pproach to OGE Hover*		Sec Approx ≤120-0	
Becord#:	_			Record #:	pproach to OGE Hover	200	3120-0	0-10
E. Takeoff Minimum Power	_				proach to OGE Hover*	>50	≤120-0	- 4.09
E. Takeon Minimum Power Record#:	-			Record#:	proach to UGE Hover	>50	5120-0	210
	38 Perform Hov	oring Eli	abt		pproach to IGE Hover*	3	≤120-0	
F. Hovering Stationary IGE*	3		aur	Record#:	pproach to IGE Hover	<u> </u>	5120-0	
Record#:		0			proach to IGE Hover*	3	≤120-0	
G. Hovering Taxi IGE*	3	0			proach to IGE Hover	3	5120-0	
Record #:				Record#:				
Record #: H. Hover OGE*	50<10K	0		W. NOE*	TASK 2026 Perform Terrain Flight		Int	
Record #:	50<10K	0		Record #:		0-25	≤120	
Record #: I. Transverse Flow [*]	_			Record #:				
I. Transverse Flow [*] Record #:	_							
	0	0				<u> </u>	<u> </u>	
J. Landing	Perform VMC FI							
K. Climb		65-80	euvers			-	-	1
Record #:	SIDK	00-00					<u> </u>	
Record #: L. Level Flight*	<10K	80					<u> </u>	
Record#:	STOK	- 00				-	<u> </u>	
Record#: M. Level Flight*	<10K	100						
M. Level Flight" Record#:	STOK	100					<u> </u>	
N. Level Flight*	<10K	120					<u> </u>	-
N. Level Flight	<10K	120						
N2. Level Flight*	<10K	145					<u> </u>	
N2. Level Flight*	STOK	149					<u> </u>	
Record #: O. Std Rate Turn	<10K	≤120					L	
O. Stol Rate Turn Record #:	<10K	5120						
Record #:								



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Methods – Data Processing

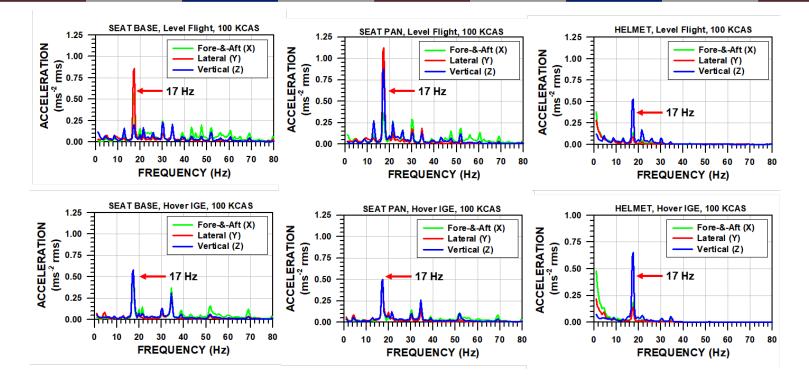
- Unweighted Acceleration Spectra (1–150 Hz)
 - Identify the frequency location/magnitude of major peaks.
 - PRF: Propeller Rotation Frequency
 - BPF: Blade Passage Frequency
- Weighted Overall Accelerations (1–80 Hz, ISO 2631-1)
 - Assess the comfort reaction and health risk.
 - pVTV: point vibration total value (vector sum of overall accelerations for three directions at seat pan and seat back)
 - oVTV: overall vibration total value (vector sum of pan and back pVTVs)
 - Estimate allowable exposure duration associated with no health risk



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Results – Unweighted Acceleration Spectra

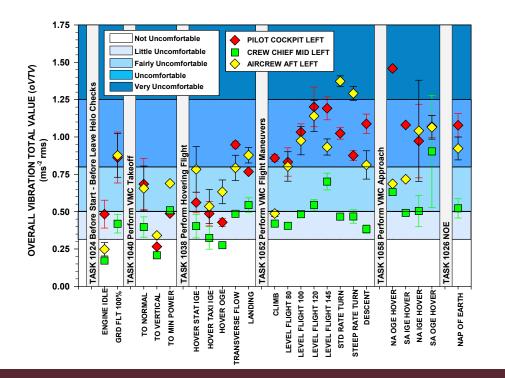


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Results – Comfort Reactions (ISO 2631-1)

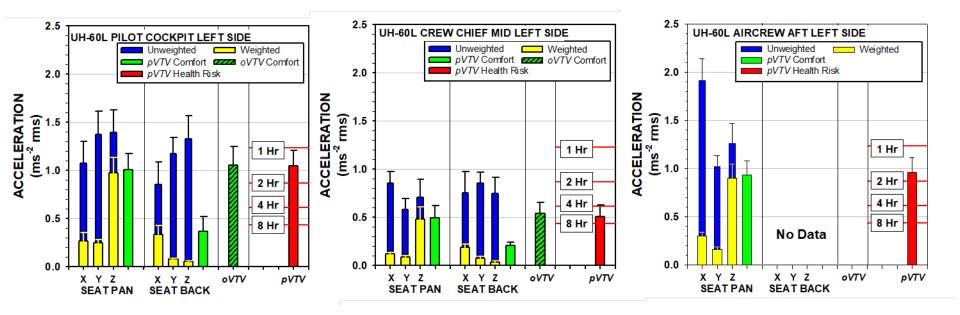


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Results – Unweighted/Weighted Overall Acceleration

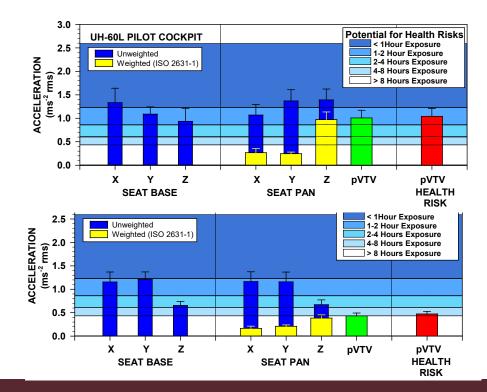




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Results – Blackhawk Comparison

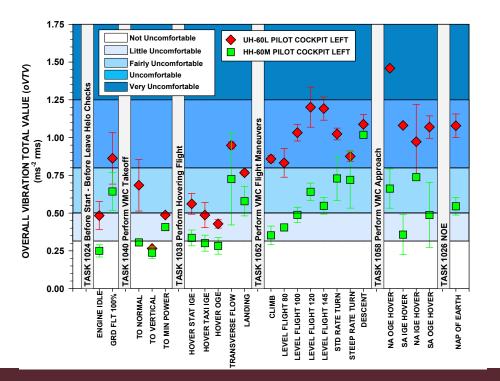




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Results – Blackhawk Comparison (Comfort)

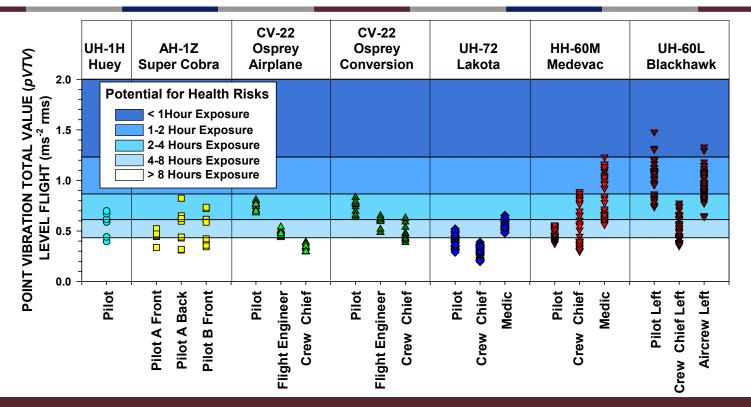


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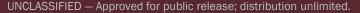
Results – Aircraft Level Flight Comparison



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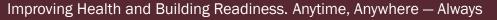




Conclusions

- Aircrew are exposed to significant higher-frequency multi-axis vibration (above 10 Hz).
- The assessment shows aircrew comfort and potential for health risks can occur in as little as 1-2 hours due to exposures over current recognized threshold limits (ACGIH, MIL-STD).
- Active aircraft vibration mitigation technologies can sufficiently reduce this threat.
- The mechanism(s) by which vibration may affect aircrew physiology and health risk are still not clear and further research is needed to improving mitigation concepts.
- The vibration data collected will be used to establish appropriate criteria for developing effective mitigation concepts through modeling.

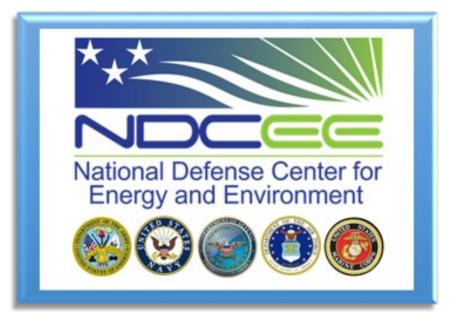






Acknowledgement

DCPH-A and AFRL would like to thank the NDCEE for their support





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